

Control of Bending Vibrations within Subdomains of Thin Plates by Piezoelectric Actuation

MICHAEL KROMMER

TECHNICAL MECHANICS



Mechatronics Group
Johannes Kepler University Linz



*Johann Radon Institute for
Computational and Applied Mathematics*

***DIRECT and INVERSE PROBLEMS
in PIEZOELECTRICITY***

Miniworkshop

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is gratefully acknowledged.

INTRODUCTION

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LINZ CENTER OF COMPETENCE IN MECHATRONICS

Krommer, M., Varadan, V.V.: **Control of Bending Vibrations within Sub – Domains of Thin plates; Part I: Theory and Exact Solution.** Journal of Applied Mechanics **72**(3), 2005, pp. 432-444.

PART I

Krommer, M.: **The Significance of Non-Local Constitutive Relations for Composite Thin Plates Including Piezoelectric Layers with Prescribed Electric Charge.** Smart Materials and Structures **12**, 2003, pp. 318-330.

PART II

Krommer, M., Varadan, V.V.: **Control of Bending Vibrations within Sub – Domains of Thin plates; Part II: Piezoelectric Actuation and Approximate Solution.** Scheduled for publication January 2006: Journal of Applied Mechanics.

PART III

*Control of Bending Vibrations within Subdomains
of Thin Plates by Piezoelectric Actuation*

PART I: Bending Vibrations of Thin Plates

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$$A: \operatorname{div}(\operatorname{div} \mathbf{M}) + p_z = P \ddot{w}_0$$

$$C: \left[\operatorname{div} \mathbf{M} \cdot \mathbf{n} + \nabla (\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} - \bar{q} \right] \delta w_0 = 0, \quad (\mathbf{M} \mathbf{n} \cdot \mathbf{n} - \bar{m}) (\nabla \delta w_0 \cdot \mathbf{n}) = 0$$

$$P: (\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \delta w_0 \Big|_{P^+}^{P^-} = 0$$

\mathbf{M} ... bending moment tensor

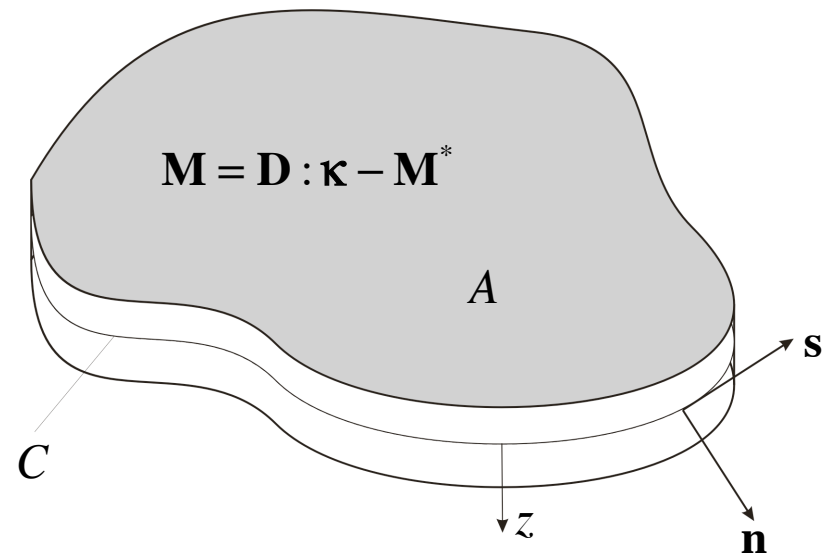
\mathbf{M}^* ... self moment tensor

$$\boldsymbol{\kappa} - \frac{1}{2} \left[\nabla \nabla w_0 + (\nabla \nabla w_0)^T \right]$$

... curvature tensor

w_0 ... deflection

p_z ... transverse force loading



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PART I:

Convolution integral formulation

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$$\int_A \hat{p}_z^d \hat{w}_0 dA - \int_{C_\psi} (\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{n}) \hat{\psi} dC + \int_{C_w} \left[\text{div} \hat{\mathbf{M}}^d \cdot \mathbf{n} + \nabla (\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} \right] \hat{w}_0 dC =$$

$$= \int_A \hat{p}_z \hat{w}_0^d dA + \int_A \hat{\mathbf{M}}^* : \hat{\mathbf{k}}^d dA - \int_{C_m} \hat{m} (\nabla \hat{w}_0^d \cdot \mathbf{n}) dC + \int_{C_q} \hat{q} \hat{w}_0^d dC + \sum_i \left[\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{s} \hat{w}_0 \right]_{P_{wi}^+}^{P_{wi}^-}$$

- $\hat{p}_z = \hat{p}_z + P(sw_0(t=0) + \dot{w}_0(t=0))$... accounts for initial conditions

- "dummy" loading case: $\left\{ \begin{array}{l} \text{homogenous initial conditions} \\ \text{homogenous kinematical and dynamical bc's} \\ \text{no self - moment applied: } \mathbf{M}^{*d} = \mathbf{0} \end{array} \right.$

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PART I:

Convolution integral formulation

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$$\int_A \hat{p}_z^d \hat{w}_0 dA - \int_{C_\psi} (\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{n}) \hat{\psi} dC + \int_{C_w} \left[\operatorname{div} \hat{\mathbf{M}}^d \cdot \mathbf{n} + \nabla (\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} \right] \hat{w}_0 dC =$$

$$= \int_A \hat{p}_z \hat{w}_0^d dA + \int_A \hat{\mathbf{M}}^* : \hat{\boldsymbol{\kappa}}^d dA - \int_{C_m} \hat{m} (\nabla \hat{w}_0^d \cdot \mathbf{n}) dC + \int_{C_q} \hat{q} \hat{w}_0^d dC + \sum_i \left[\hat{\mathbf{M}}^d \mathbf{n} \cdot \mathbf{s} \hat{w}_0 \right]_{P_{wi}^+}^{P_{wi}^-}$$

- kinematical bc's ...
$$\begin{cases} C = C_w \cup C_q, & C = C_\psi \cup C_m \\ C_w : w_0 = \bar{w}_0, & C_\psi : \nabla w_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$
- C_w and C_ψ such that, if $\bar{w}_0 = 0$ and $\bar{\psi} = 0$, then no rigid body motion can occur!
- Decomposition of deflection, $w_0 = \check{w}_0 + \tilde{w}_0$, with:
$$\begin{cases} C_w : \check{w}_0 = \bar{w}_0 \\ C_\psi : \nabla \check{w}_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$

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Convolution integral formulation

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$$\int_A \hat{p}_z^d \hat{w}_0^d dA = \int_A \tilde{p}_z \hat{w}_0^d dA + \int_A \hat{\mathbf{M}}^* : \hat{\boldsymbol{\kappa}}^d dA - \int_{C_m} \hat{m} (\nabla \hat{w}_0^d \cdot \mathbf{n}) dC + \int_{C_q} \hat{q} \hat{w}_0^d dC$$

- A : $\tilde{p}_z = \hat{p}_z + P(sw_0(t=0) + \dot{w}_0(t=0)) - Ps^2 \hat{w}_0 + \text{div}(\text{div} \hat{\mathbf{M}})$

- $\left\{ \begin{array}{l} C_m : \hat{m} = \tilde{m} - \hat{\mathbf{M}} \mathbf{n} \cdot \mathbf{n} \\ C_q : \hat{q} = \tilde{q} - \text{div} \hat{\mathbf{M}} \cdot \mathbf{n} - \nabla (\hat{\mathbf{M}} \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} \end{array} \right\}$ with $\tilde{\mathbf{M}} = \mathbf{D} : \tilde{\boldsymbol{\kappa}}$, $\tilde{\boldsymbol{\kappa}} = -\frac{1}{2} [\nabla \nabla \tilde{w}_0 + (\nabla \nabla \tilde{w}_0)^T]$

- statically admissible moment tensor \mathbf{M}^p : $\left\{ \begin{array}{l} A : \text{div}(\text{div} \mathbf{M}^p) + \tilde{p}_z = 0 \\ C_q : \text{div} \mathbf{M}^p \cdot \mathbf{n} + \nabla (\mathbf{M}^p \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} = \tilde{q} \\ C_m : \mathbf{M}^p \mathbf{n} \cdot \mathbf{n} = \tilde{m}, \bar{P}_i : [\mathbf{M}^p \mathbf{n} \cdot \mathbf{s}]_{\bar{P}_i^+}^{\bar{P}_i^-} = 0 \end{array} \right.$

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PART I:

Convolution integral formulation

$$\int_0^t \int_A p_z^d(\mathbf{x}, \tau) \tilde{w}_0(\mathbf{x}, t - \tau) dA d\tau = \int_0^t \int_A [\mathbf{M}^p(\mathbf{x}, \tau) + \mathbf{M}^*(\mathbf{x}, t - \tau)] : \boldsymbol{\kappa}^d dA d\tau$$

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Due to the arbitrariness of p_z^d the deflection $\tilde{w}_0 = w_0 - \tilde{w}_0$ vanishes identically, if $\mathbf{M}^p + \mathbf{M}^* = \mathbf{0}$. Therefore, applying control by means of a self moment can be used to eliminate the bending motion \tilde{w}_0 . The motion \tilde{w}_0 , which we account to the non homogenous kinematical bc's is not eliminated.

Hence, $w_0 = \tilde{w}_0$.

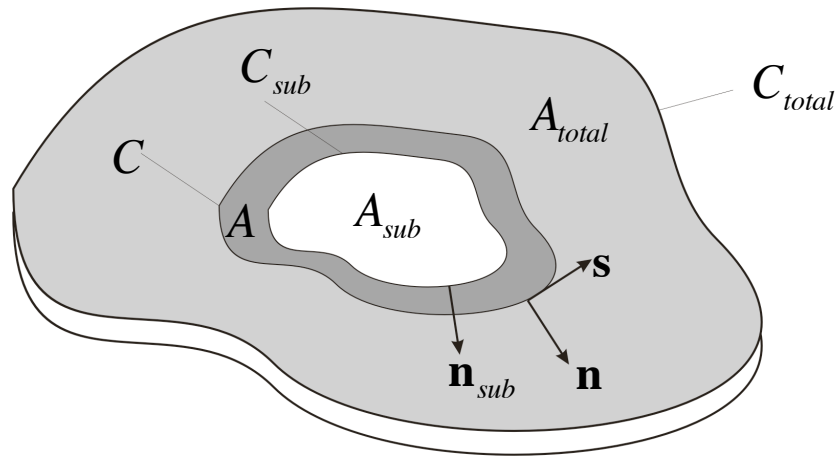
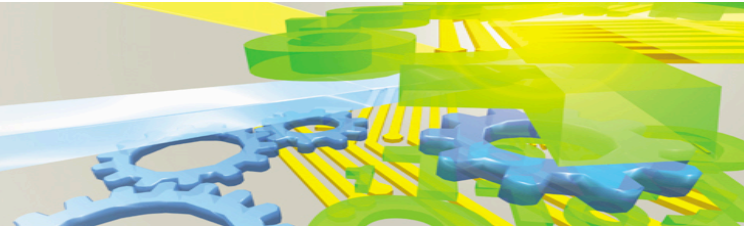
- statically admissible moment tensor $\mathbf{M}^p = m^p \mathbf{I}$:

$$\begin{cases} A : \Delta m^p + \tilde{p}_z = 0 \\ C_q : \nabla m^p \cdot \mathbf{n} = \tilde{q} \\ C_m : m^p = \tilde{m} \end{cases}$$

→ solution for \mathbf{M}^p exists, but may be non unique!

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PART I: Subdomain control



- Boundary conditions for free body A:

$$C : \begin{cases} w_0 = \bar{w}_0, & \mathbf{Mn} \cdot \mathbf{n} = \bar{m} \\ \text{div} \mathbf{M} \cdot \mathbf{n} + \nabla(\mathbf{Mn} \cdot \mathbf{s}) \cdot \mathbf{s} = \bar{q}, & \nabla w_0 \cdot \mathbf{n} = \bar{\psi} \\ w_0 = \bar{w}_0, & \nabla w_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$

- Desired deflection \tilde{w}_0 :

$$A_{sub} : \tilde{w}_0 = 0, \quad C_{sub} : \tilde{w}_0 = 0, \quad \nabla \tilde{w}_0 \cdot \mathbf{n}_{sub} = 0$$

$$C : \begin{cases} \tilde{w}_0 = \bar{w}_0 \\ \nabla \tilde{w}_0 \cdot \mathbf{n} = \bar{\psi} \\ \tilde{w}_0 = \bar{w}_0, \quad \nabla \tilde{w}_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$

- Statically admissible moment tensor:

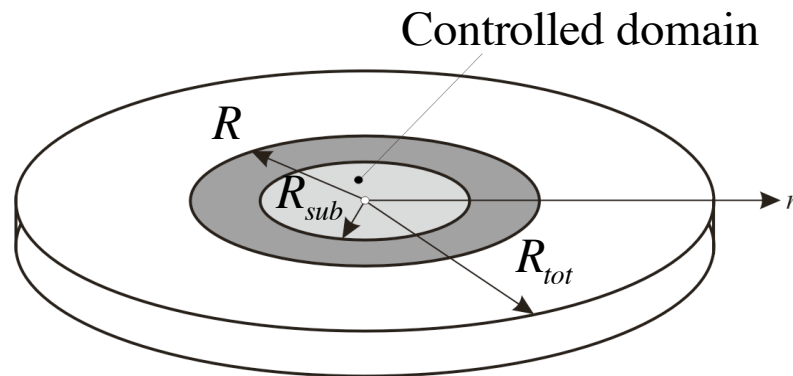
$$A : \Delta m^p + \tilde{p}_z = 0$$

$$C : \begin{cases} m^p = \tilde{m} \\ \nabla m^p \cdot \mathbf{n} = \tilde{q} \\ \text{no boundary conditions to be satisfied} \end{cases}$$



PART I:

Axisymmetric bending of circular plates



- Clamped circular plate:

$$p_z(r, \varphi, t) = p_0(t)$$

- Boundary conditions for free subdomain $r \leq R$:

$$w_0 = \bar{w}_0, \quad \partial w_0 / \partial r = \bar{\psi}$$

- Desired deflection \check{w}_0 :

$$r \leq R_{sub} : \check{w}_0 = 0$$

$$R_{sub} \leq r \leq R : \check{w}_0 = C_1 + C_2 r^2 + C_3 r^4 + C_4 r^6 + C_5 r^8 + C_6 r^{10}$$

$$r = R_{sub} : \check{w}_0 = 0, \quad \frac{\partial \check{w}_0}{\partial r} = 0 \quad \text{and} \quad r = R : \check{w}_0 = \bar{w}_0, \quad \frac{\partial \check{w}_0}{\partial r} = \bar{\psi}$$

\check{w}_0 is adjusted to the four conditions; hence four unknowns remain!

PART I:

Axisymmetric bending of circular plates

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- Statically admissible moment:

$$r \leq R: \frac{1}{r} \frac{\partial m^p}{\partial r} + \frac{\partial^2 m^p}{\partial r^2} + \tilde{p}_0 = 0$$

$$r \leq R_{sub}: \tilde{p}_0 = p_0, \quad R_{sub} \leq r \leq R: \tilde{p}_0 = p_0 - P\ddot{w}_0 - D\Delta\Delta\tilde{w}_0$$

No bc for m^p ; hence, one more unknown!

If $\mathbf{M}^* = -\mathbf{M}^p = -m^p \mathbf{I}$ then the motion for $r \leq R_{sub}$ vanishes;
for $R_{sub} \leq r \leq R$ the motion is \tilde{w}_0 , as it has been specified in the series.

Five unknowns present in the solution!

$\bar{w}_0, \bar{\psi}$, m^p is not unique, 2 constants from power series

- 2 from to be admissible
- 2 from post calculation: $\bar{w}_0, \bar{\psi}$ are deflection and slope of controlled plate at $r = R$
- 1 remains as additional degree of freedom

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PART I:

Numerical results

- Parameters:

$$D = 162.45 \text{ Nm}, \quad P = 8.1 \text{ kgm}^{-2}, \quad p_0(t) = P_0 e^{i\omega t}, \quad P_0 = 100 \text{ Nm}^{-2}$$

- Geometry: $R_{tot} = 0.2 \text{ m}, \quad R = 0.04 \text{ m}, \quad R_{sub} = 0.03 \text{ m}$

- Natural frequencies: $f_1 = 182.0 \text{ Hz}, \quad f_2 = 708.7 \text{ Hz}, \quad f_3 = 1588 \text{ Hz}$

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$$w_0(r, t) = W(r) e^{i\omega t}, \quad \tilde{w}_0(r, t) = \tilde{W}_0(r) e^{i\omega t}$$

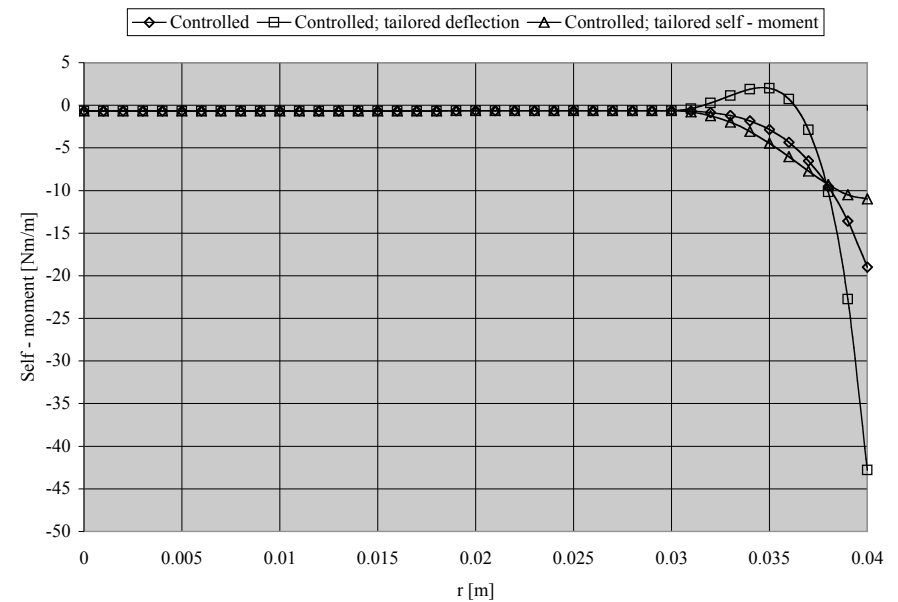
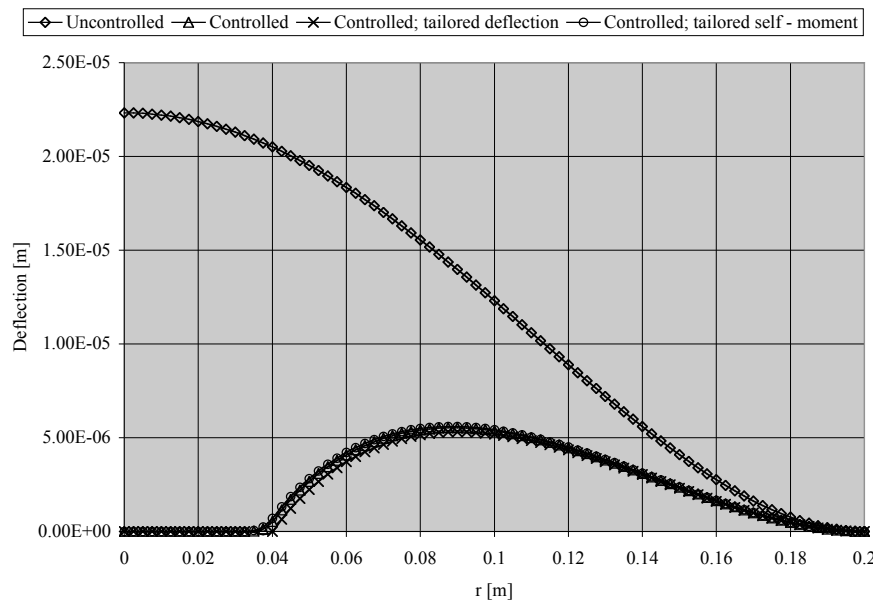
$$D\Delta\Delta W(r) - P\omega^2 W(r) = \begin{cases} 0 \leq r \leq R_{sub} : & = 0 \\ R_{sub} \leq r \leq R : & = -P\omega^2 \tilde{W}_0 + D\Delta\Delta \tilde{W}_0(r) \\ R \leq r \leq R_{tot} : & = P_0 \end{cases}, \quad r = R_{tot} : \quad W = 0, \quad \frac{\partial W}{\partial r} = 0$$

$$W(r = R) e^{i\omega t} = \bar{w}_0 = \bar{W}_0 e^{i\omega t}, \quad \frac{\partial W}{\partial r}(r = R) e^{i\omega t} = \bar{\psi} = \bar{\Psi} e^{i\omega t}$$

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PART I:

Numerical results



Deflection and self-moment for clamped plate: $\omega = 2\pi 100 s^{-1}$

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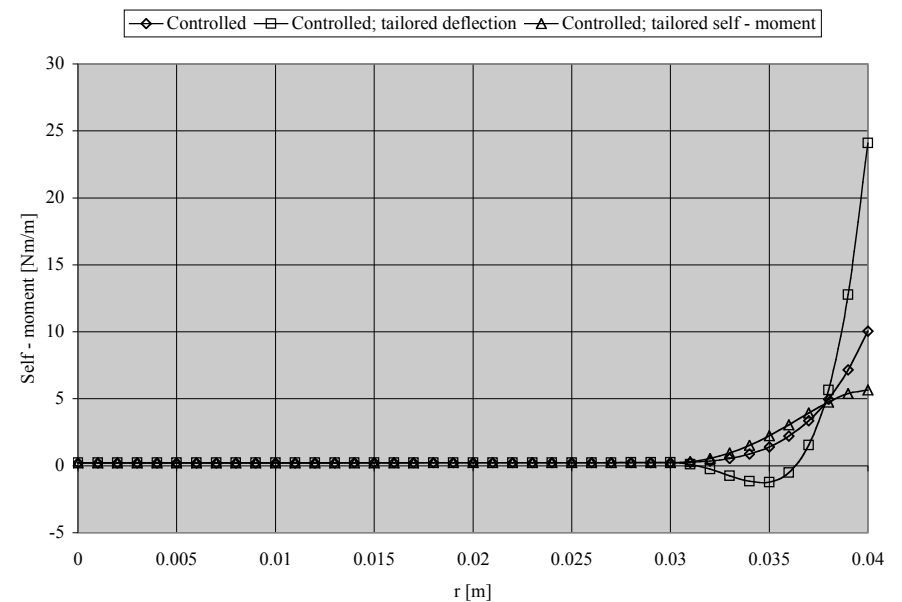
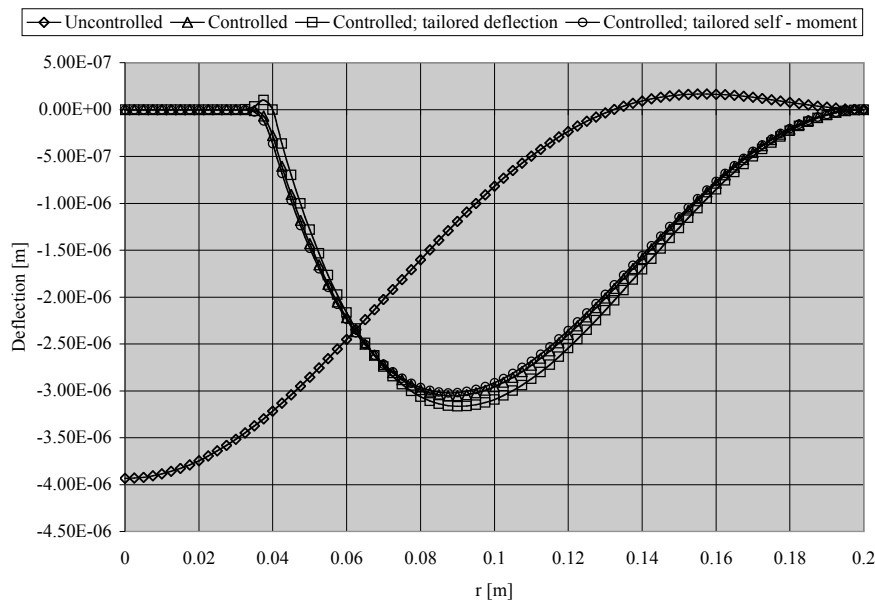
PART I:

Numerical results

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Deflection and self-moment for clamped plate: $\omega = 2\pi 500 s^{-1}$

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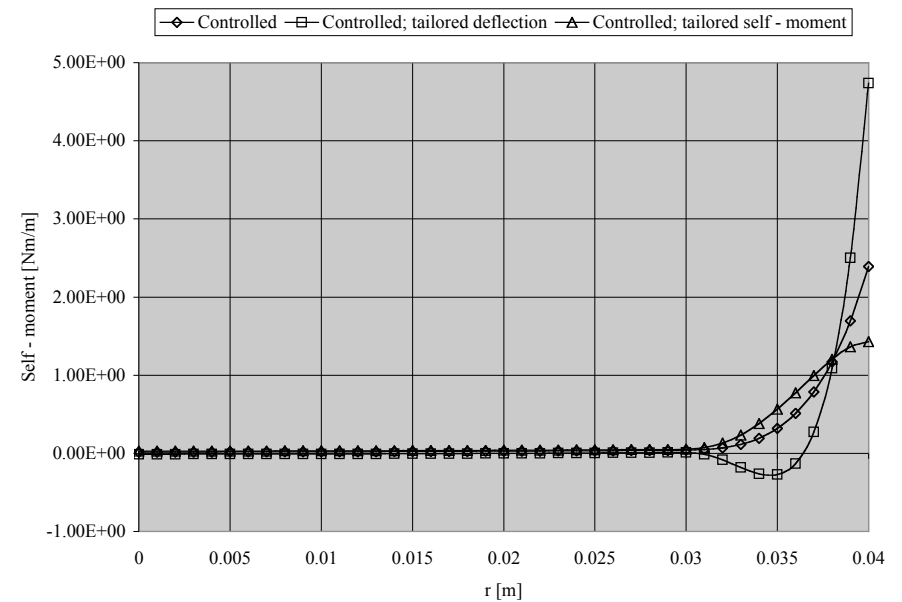
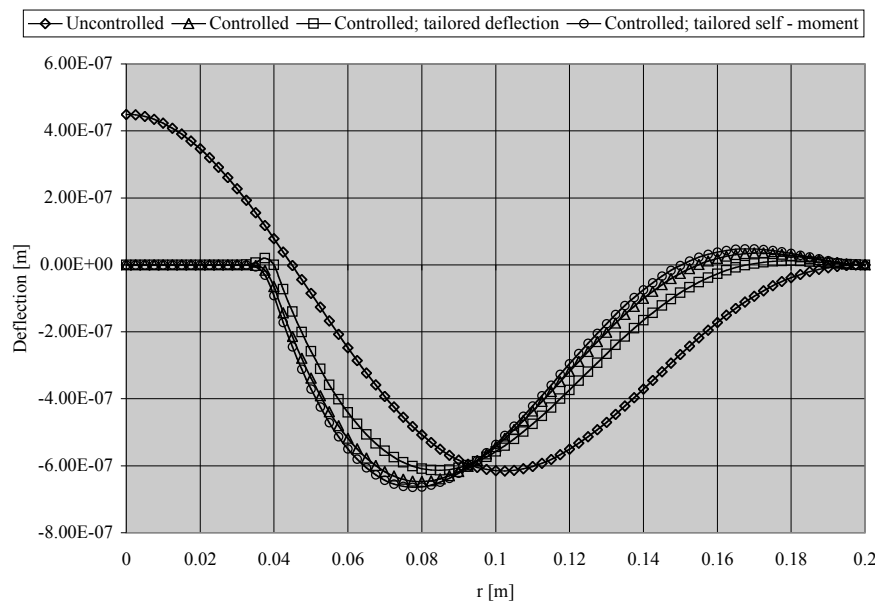
PART I:

Numerical results

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Deflection and self-moment for clamped plate: $\omega = 2\pi 1000 s^{-1}$

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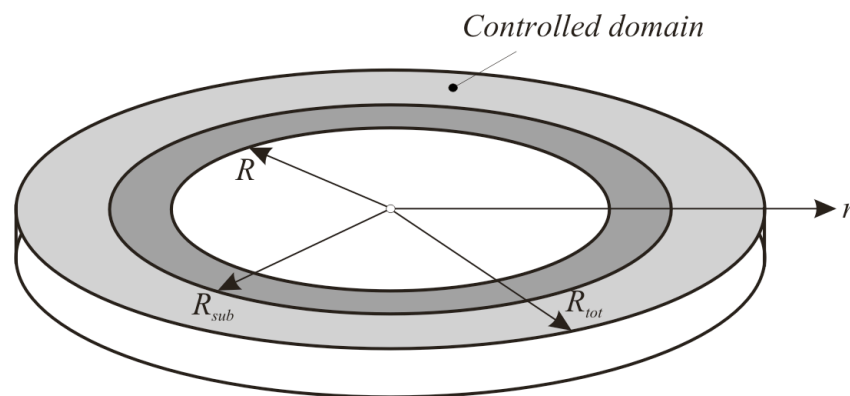
PART I:

Numerical results; control of alternative subdomains

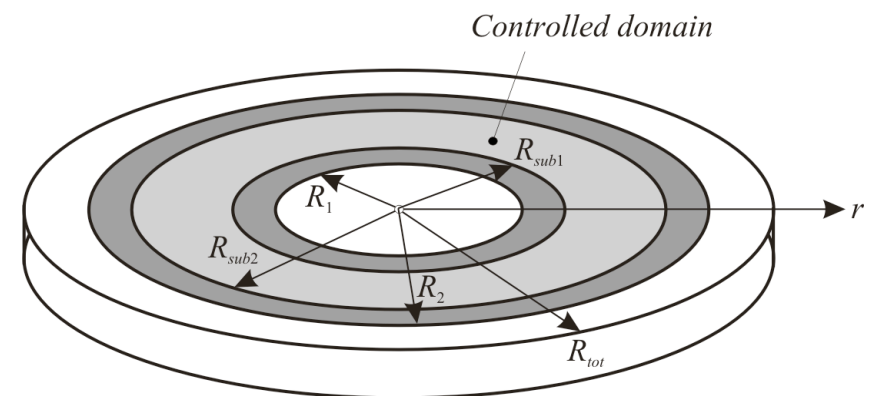
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$$R_{tot} = 0.2m, \quad R = 0.16m, \quad R_{sub} = 0.17m$$



$$R_1 = 0.075m, \quad R_{sub1} = 0.085m$$
$$R_{sub2} = 0.115m, \quad R_2 = 0.125m$$

***Control of Bending Vibrations within Subdomains
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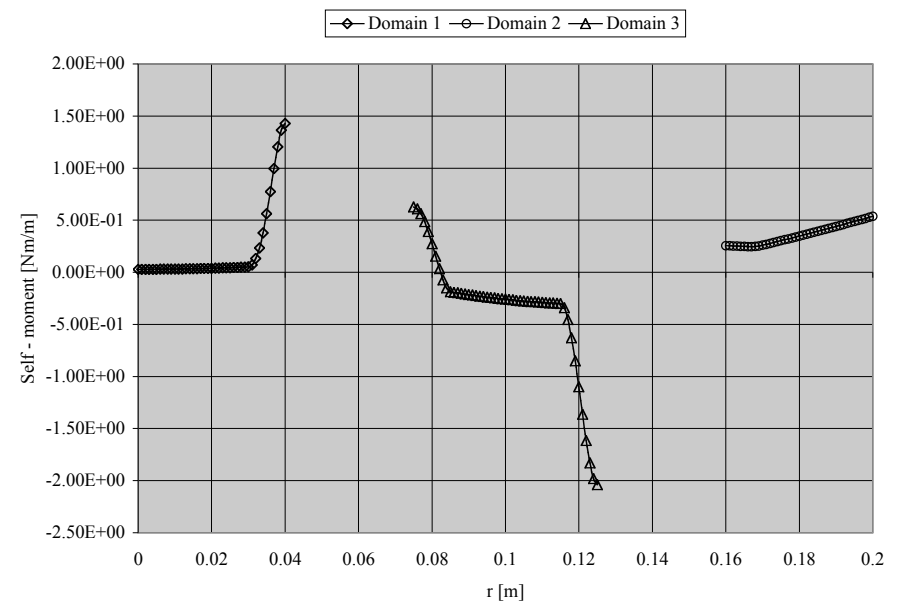
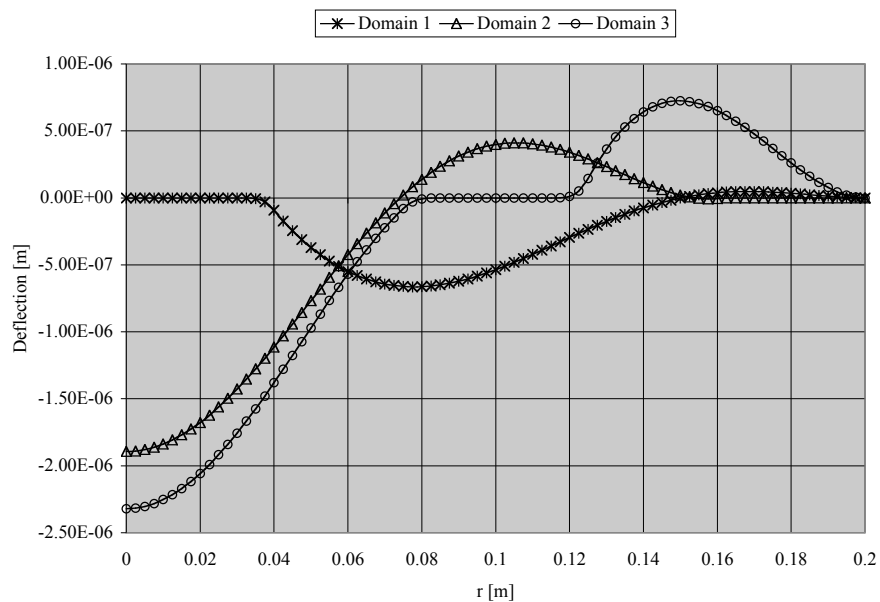
PART I:

Numerical results; control of alternative subdomains

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Deflection and self-moment for clamped plate: $\omega = 2\pi 1000 s^{-1}$

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PART II: Governing equations; mechanical

$$A: \operatorname{div}(\operatorname{div} \mathbf{M}) + p_z = P \ddot{w}_0$$

$$C: [\operatorname{div} \mathbf{M} \cdot \mathbf{n} + \nabla(\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} - \bar{q}] \delta w_0 = 0, \quad (\mathbf{M} \mathbf{n} \cdot \mathbf{n} - \bar{m})(\nabla \delta w_0 \cdot \mathbf{n}) = 0$$

$$P: (\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \delta w_0 \Big|_{P^+}^{P^-} = 0$$

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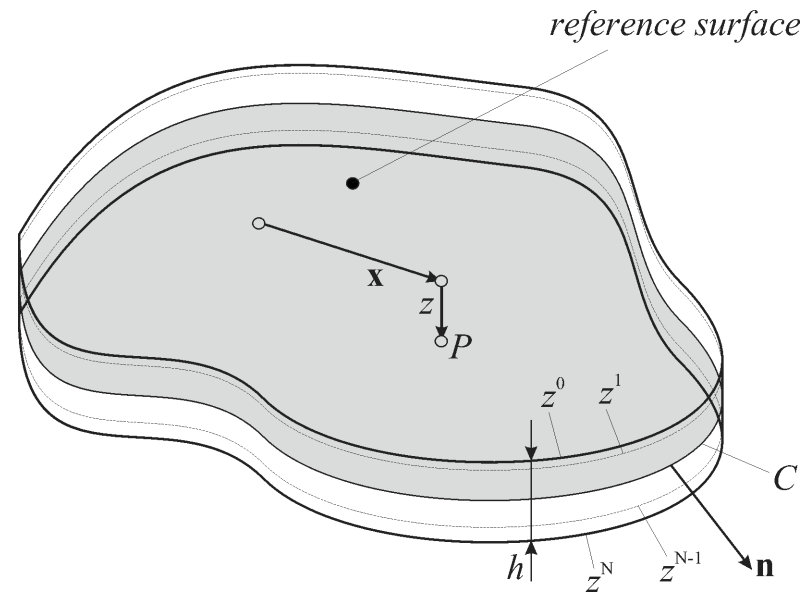


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$$\mathbf{M} = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \boldsymbol{\sigma}^k z dz, \quad P = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \rho^k dz$$

$\boldsymbol{\sigma}^k$... plane stress tensor

ρ^k ... density



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PART II: *Constitutive relation; mechanical*

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$$\boldsymbol{\sigma}^k = \mathbf{C}^k z : \boldsymbol{\kappa} - \mathbf{e}^k E_z^k = \mathbf{C}^k : (z \boldsymbol{\kappa} - \boldsymbol{\varepsilon}^{*k}), \quad \boldsymbol{\varepsilon}^{*k} \dots \text{eigenstrain}$$

\mathbf{C}^k ... tensor of elastic moduli

\mathbf{e}^k ... tensor of piezoelectric coefficients

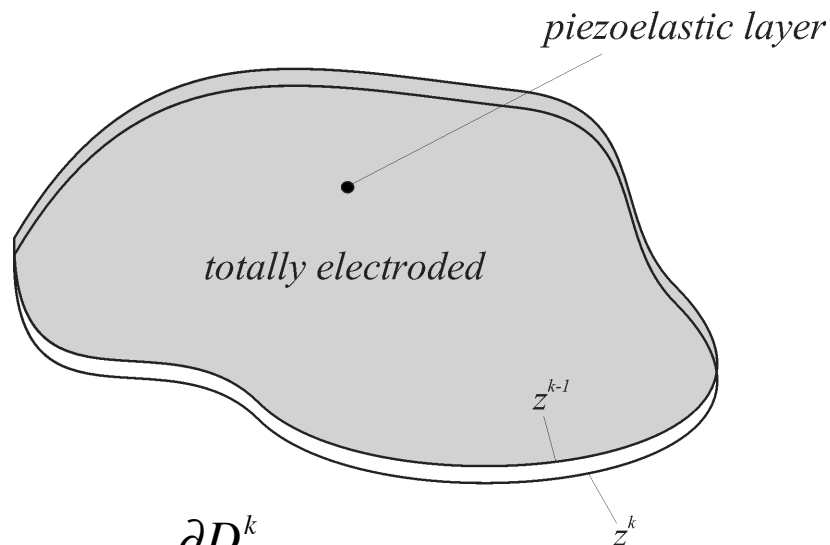
$E_z^k = -\nabla \phi \cdot \mathbf{e}_z$... thickness component of electric field vector

$$\mathbf{M} = \mathbf{D} : \boldsymbol{\kappa} - \mathbf{M}^*$$

$$\mathbf{M} = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \boldsymbol{\sigma}^k z dz = \underbrace{\left(\sum_{k=1}^N \int_{z^{k-1}}^{z^k} \mathbf{C}^k z^2 dz \right)}_{\mathbf{D}} : \boldsymbol{\kappa} - \underbrace{\sum_{k=1}^N \int_{z^{k-1}}^{z^k} \mathbf{e}^k E_z^k z dz}_{\mathbf{M}^*}$$

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PART II: Governing equations; electrostatic



$$D_z^k = \mathbf{e}^k : z \mathbf{\kappa}^k + \eta^k E_z^k = \eta^k (E_z^k - E_z^{*k})$$

η^k ... electric permittivity

$$E_z^{*k} = \frac{\mathbf{e}^k : z \mathbf{\kappa}^k}{\eta^k} \text{ ... electric eigenfield}$$

$$\operatorname{div} \mathbf{D}^k + \frac{\partial D_z^k}{\partial z} = 0$$

\mathbf{D}^k ... in - plane electric displacement vector

D_z^k ... thickness component of electric displacement vector

→ due to the thinness of the layer we neglect \mathbf{D}^k

PART II: Governing equations; electrostatic



$$\frac{\partial D_z^k}{\partial z} = 0, \quad D_z^k = \eta^k (E_z^k - E_z^{*k})$$

$$D_z^k = \text{const.} = \frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} D_z^k dz = \eta^k \frac{V^k}{z^k - z^{k-1}} - \frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} \eta^k E_z^{*k} dz$$

- Relation between Q^k and V^k from Gauss law of electrostatics:

$$Q^k = \int_A D_z^k dA = A \eta^k \frac{V^k}{z^k - z^{k-1}} - \int_A \frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} \eta^k E_z^{*k} dz dA$$

$$E_z^k = E_z^{k,c} + E_z^{*k} - \left[\frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} E_z^{*k} dz - \kappa_Q^k \frac{1}{A} \int_A \frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} E_z^{*k} dz dA \right]$$

$$E_z^{k,c} = \kappa_Q^k \frac{Q^k}{A \eta^k} + (1 - \kappa_Q^k) \frac{V^k}{h^k}$$

PART II: Effective constitutive relations

$$\mathbf{M} = \mathbf{D} : \boldsymbol{\kappa} - \mathbf{M}^*$$

$$\mathbf{M}^* = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \mathbf{e}^k E_z^k z dz, \quad E_z^k = E_z^{k,c} + E_z^{*k} - \left[\frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} E_z^{*k} dz - \boldsymbol{\kappa}_Q^k \frac{1}{A} \int_A \frac{1}{z^k - z^{k-1}} \int_{z^{k-1}}^{z^k} E_z^{*k} dz dA \right]$$

- Reformulation of constitutive relations: $\mathbf{M} = \mathbf{D}^{eff} : \boldsymbol{\kappa} + \mathbf{d} : \bar{\boldsymbol{\kappa}} - \mathbf{M}^{*,eff}$

$$\mathbf{D}^{eff} = \mathbf{D} + \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \frac{\mathbf{e}^k \otimes \mathbf{e}^k}{\eta^k} \left(z - \frac{1}{2} (z^k + z^{k-1}) \right) z dz, \quad \mathbf{d} = \sum_{k=1}^N \boldsymbol{\kappa}_Q^k \int_{z^{k-1}}^{z^k} \frac{\mathbf{e}^k \otimes \mathbf{e}^k}{\eta^k} \frac{1}{2} (z^k + z^{k-1}) z dz$$

$$\bar{\boldsymbol{\kappa}} = \frac{1}{A} \int_A \boldsymbol{\kappa} dA \quad \dots \text{mean curvature tensor}$$

$$\mathbf{M}^{*,eff} = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \mathbf{e}^k E_z^{k,c} z dz \quad \dots \text{effective piezoelectric (self-) moment tensor}$$

PART II:

Dynamic analysis of simply supported plates

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- Transversally isotropic materials
- Isotropic plane parallel to reference surface

$$\mathbf{M} = D^{eff} \nu^{eff} \mathbf{I} tr \boldsymbol{\kappa} + D^{eff} (1 - \nu^{eff}) \boldsymbol{\kappa} + d \mathbf{I} tr \bar{\boldsymbol{\kappa}} - m^{*,eff} \mathbf{I}$$

$$D^{eff} = D + D^{me}, \quad \nu^{eff} = \frac{D\nu + D^{me}}{D + D^{me}}, \quad d = \sum_{k=1}^N \kappa_Q^k \int_{z^{k-1}}^{z^k} \frac{e^k e^k}{\eta^k} \frac{1}{2} (z^k + z^{k-1}) z dz$$

$$D = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} Y^k z^2 dz, \quad \nu = \frac{1}{D} \sum_{k=1}^N \int_{z^{k-1}}^{z^k} Y^k \nu^k z^2 dz, \quad D^{me} = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} \frac{e^k e^k}{\eta^k} \left(z - \frac{1}{2} (z^k + z^{k-1}) \right) z dz$$

$$m^{*,eff} = \sum_{k=1}^N \int_{z^{k-1}}^{z^k} e^k \left(\kappa_Q^k \frac{Q^k}{A \eta^k} + (1 - \kappa_Q^k) \frac{V^k}{h^k} \right) z dz$$

- sensor equation:

$$y^k = Q^k - A \eta^k \frac{V^k}{h^k} = - \int_A e^k \Delta w_0 z_m^k dA$$

*Control of Bending Vibrations within Subdomains
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PART II:

Dynamic analysis of simply supported plates

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- Simply supported edges
- Polygonal domain

$$\Delta(D^{eff} \Delta w_0) + P\ddot{w}_0 = p_z - \Delta m^{*,eff}, \quad \text{at } C: \quad w_0 = 0, \quad D^{eff} \Delta w_0 + d \frac{1}{A} \int_A \Delta w_0 dA = -m^{*,eff}$$

- domain wise constant parameters

$$D^{eff} \Delta \Delta w_0 + P\ddot{w}_0 = p_z - D^{eff} (1 - \nu^{eff}) \Delta \kappa^{eff}, \quad \text{at } C: \quad w_0 = 0, \quad \Delta w_0 = -(1 - \nu^{eff}) \kappa^{eff}$$

$$\kappa^{eff} = \frac{m^{*,eff} + d \frac{1}{A} \int_A \Delta w_0 dA}{D^{eff} (1 - \nu^{eff})} = \frac{m^{*,eff} + m^{*,f}}{D^{eff} (1 - \nu^{eff})}$$

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- Frequency domain formulation: $p_z = \hat{p}e^{i\omega t}$, $m^{*,eff} = \hat{m}^{*,eff} e^{i\omega t}$, $w_0 = \hat{W}e^{i\omega t}$

$$D^{eff} \Delta \Delta \hat{W} - \mu \hat{W} = \hat{p} - D^{eff} (1 - \nu^{eff}) \Delta \hat{\kappa}^{eff}, \quad \text{at } C: \quad \hat{W} = 0, \quad \Delta \hat{W} = -(1 - \nu^{eff}) \hat{\kappa}^{eff}$$

$$\mu = \omega^2 P, \quad \hat{\kappa}^{eff} = \frac{\hat{m}^{*,eff} + d \frac{1}{A} \int_A \Delta \hat{W} dA}{D^{eff} (1 - \nu^{eff})} = \frac{\hat{m}^{*,eff} + \hat{m}^{*,f}}{D^{eff} (1 - \nu^{eff})}$$

- Decomposition of fourth order problem

into two second order problems of Helmholtz - Klein - Gordon type:

$$\hat{W} = \hat{W}_1 + \hat{W}_2, \quad \Delta \hat{W}_j - (-1)^j \alpha \hat{W}_j = (-1)^j \frac{\alpha}{2\mu} \hat{p} - \frac{\hat{m}^{*,eff} + \hat{m}^{*,f}}{2D^{eff}}, \quad \text{at } C_0: \quad \hat{W}_j = 0$$

$$\alpha = \sqrt{\frac{\mu}{D^{eff}}}, \quad \hat{m}^{*,f} = d \frac{1}{A} \int_A \Delta \hat{W} dA = d \frac{1}{A} \int_A \Delta (\hat{W}_1 + \hat{W}_2) dA$$

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- Green's function method for solution: $G_j^\alpha(\mathbf{x}, \xi)$

$$\Delta \hat{W}_j - (-1)^j \alpha \hat{W}_j = 0, \quad \text{at } C_0: \quad \hat{W}_j = 0$$

- Solution is:

$$\hat{W}_j(\xi) = -\frac{\hat{m}^{*,f}}{2D^{eff}} \int_A G_j^\alpha(\mathbf{x}, \xi) dA(\mathbf{x}) + \int_A \left((-1)^j \frac{\alpha}{2\mu} \hat{p}(\mathbf{x}) - \frac{\hat{m}^{*,eff}(\mathbf{x})}{2D^{eff}} \right) G_j^\alpha(\mathbf{x}, \xi) dA(\mathbf{x})$$

- Integrate solution with respect to plate area A :

$$\begin{aligned} & \hat{m}^{*,f} \left(1 - \frac{\alpha}{2A} \frac{d}{D^{eff} + d} \int_A \left[\int_A (G_1^\alpha(\mathbf{x}, \xi) - G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) \right] dA(\xi) \right) = \\ & = \frac{1}{2} \frac{d}{D^{eff} + d} \frac{1}{A} \left(\int_A \left[\int_A \hat{p}(\mathbf{x}) (G_1^\alpha(\mathbf{x}, \xi) + G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) + \right. \right. \\ & \left. \left. + \int_A \alpha \hat{m}^{*,eff}(\mathbf{x}) (G_1^\alpha(\mathbf{x}, \xi) - G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) - 2\hat{m}^{*,eff}(\xi) \right] dA(\xi) \right) \end{aligned}$$

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FREE VIBRATIONS (for $\hat{m}^{*,f} = 0$ classical problem)

- characteristic equation for $\hat{m}^{*,f} \neq 0$:

$$2A = \alpha \frac{d}{D^{eff} + d} \int_A \left[\int_A (G_1^\alpha(\mathbf{x}, \xi) - G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) \right] dA(\xi), \quad \omega_i = \alpha_i \sqrt{\frac{D^{eff}}{P}}$$

- Eigenfunctions:

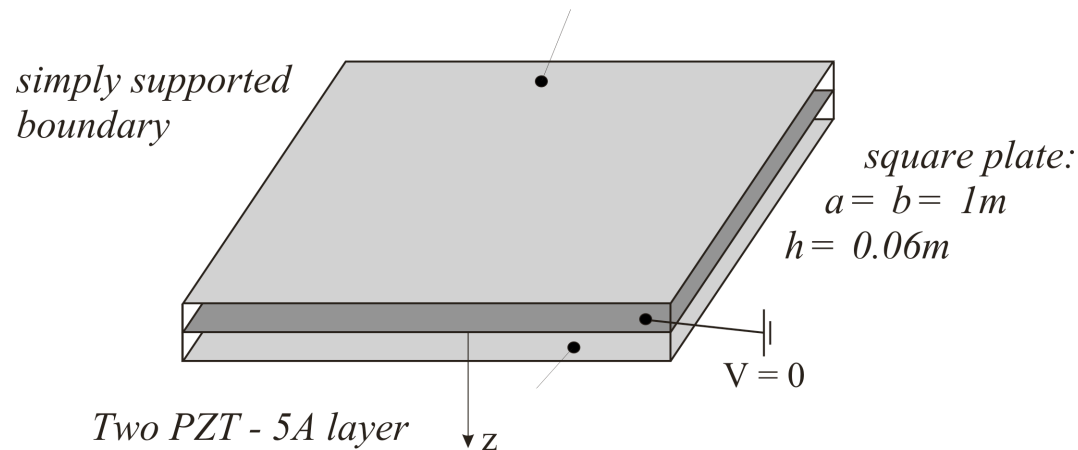
$$\hat{W}^{\alpha_i}(\xi) = \int_A (G_1^{\alpha_i}(\mathbf{x}, \xi) + G_2^{\alpha_i}(\mathbf{x}, \xi)) dA(\mathbf{x})$$

FORCED VIBRATIONS

$$y^k = -e^k z_m^k \left\{ \int_A \left[\alpha (\hat{W}_2(\xi) - \hat{W}_1(\xi)) - \frac{\hat{m}^{*,eff}(\xi)}{D^{eff}} \right] dA(\xi) - \frac{\hat{m}^{*,f}}{D^{eff}} A \right\}$$

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PART II: ***Numerical example***



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- **SQUARE PLATE:** $a = b = 1m$ and $h = 0.06m$
- **TWO** identical PZT-5A layers: interface is grounded and outer sides are electroded
- Electrodes either short - circuited or left open
- $p_z(\mathbf{x}, t) = \hat{p}(\mathbf{x})e^{i\omega t} = p_0 e^{i\omega t}$, $m^{*,eff} = 0$

***Control of Bending Vibrations within Subdomains
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PART II: Numerical example

- FREE VIBRATIONS:

- $m^{*,f} = 0$: Short - circuited electrodes or either n or m are even

$$\omega_{mn} = \alpha_{mn} \sqrt{\frac{D^{eff}}{P}} = \left[\left(\frac{m\pi}{a} \right)^2 + \left(\frac{n\pi}{a} \right)^2 \right] \sqrt{\frac{D^{eff}}{P}}, \quad \hat{W}_{mn}(x, y) = \sin\left(\frac{m\pi}{a}x\right) \sin\left(\frac{n\pi}{a}y\right)$$

- $m^{*,f} \neq 0$: Electrodes left open and n and m odd

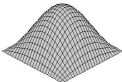
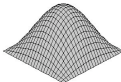
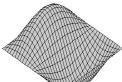
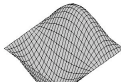
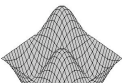
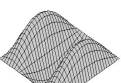
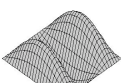
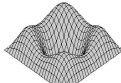
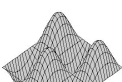
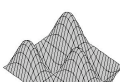
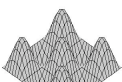
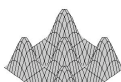
$$2A = \alpha_{mn} \frac{d}{D^{eff} + d} \int_A \left[\int_A (G_1^{\alpha_{mn}}(\mathbf{x}, \xi) - G_2^{\alpha_{mn}}(\mathbf{x}, \xi)) dA(\mathbf{x}) \right] dA(\xi), \quad \hat{W}_{mn}(\xi) = \int_A (G_1^{\alpha_{mn}}(\mathbf{x}, \xi) + G_2^{\alpha_{mn}}(\mathbf{x}, \xi)) dA(\mathbf{x})$$

$$\int_A G_j^{\alpha_{mn}}(\mathbf{x}, \xi) dA(\mathbf{x}) = \sum_{k=1,3,\dots}^{\infty} \sum_{l=1,3,\dots}^{\infty} \frac{16}{kl\pi^2} \frac{1}{-(-1)^j \alpha_{mn} - \left[\left(\frac{k\pi}{a} \right)^2 + \left(\frac{l\pi}{a} \right)^2 \right]} \sin\left(\frac{k\pi}{a}\xi\right) \sin\left(\frac{l\pi}{a}\eta\right)$$

$$j = 1, 2, \quad 0 \leq x \leq a, \quad 0 \leq y \leq a$$

PART II:

Numerical example

<p>(1,1)</p>  <p>SPC: 1027.909872 SHORT: 1055.245955</p>	<p>(1,1)</p>  <p>OPEN: 1104.565173</p>	<p>(1,2) and (2,1)</p>   <p>SPC: 2569.774679 SHORT: 2638.114889 OPEN: 2638.114889</p>
<p>(2,2)</p>  <p>SPC: 4111.639487 SHORT: 4220.983823 OPEN: 4220.983823</p>	<p>(1,3) and (3,1)</p>   <p>SPC: 5139.549359 SHORT: 5276.229779</p>	<p>(1,3) = (3,1)</p>  <p>OPEN: 5333.380586</p>
<p>(2,3) and (3,2)</p>   <p>SPC: 6681.414167 SHORT: 6859.098713 OPEN: 6859.098713</p>	<p>(3,3)</p>  <p>SPC: 9251.188847 SHORT: 9497.213602</p>	<p>(3,3)</p>  <p>OPEN: 9502.970179</p>

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Natural frequencies [rad/s] and
mode shapes of the square plate

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PART II: Numerical example



- FORCED VIBRATIONS:

- Sensor signal in upper layer:
$$y^u = -e z_m^u \left[\alpha \int_A (\hat{W}_2(\xi) - \hat{W}_1(\xi)) dA(\xi) - \frac{\hat{m}^{*,f}}{D^{eff}} A \right]$$

- Dynamic magnification factor:
$$\hat{y} = 20 \log |y^u / y_{SPC}^{u(qs)}|$$

- Solution is:

$$\hat{W}_j(\xi) = -\frac{\hat{m}^{*,f}}{2D^{eff}} \int_A G_j^\alpha(\mathbf{x}, \xi) dA(\mathbf{x}) + \int_A (-1)^j \frac{\alpha}{2\mu} \hat{p}(\mathbf{x}) G_j^\alpha(\mathbf{x}, \xi) dA(\mathbf{x})$$

$$\text{with } \hat{m}^{*,f} \left(1 - \frac{\alpha}{2A} \frac{d}{D^{eff} + d} \int_A \left[\int_A (G_1^\alpha(\mathbf{x}, \xi) - G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) \right] dA(\xi) \right) =$$

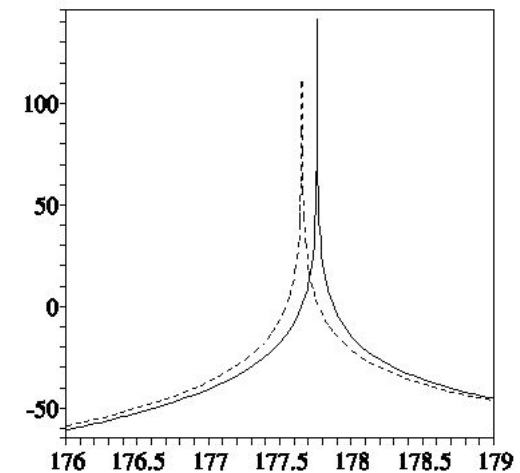
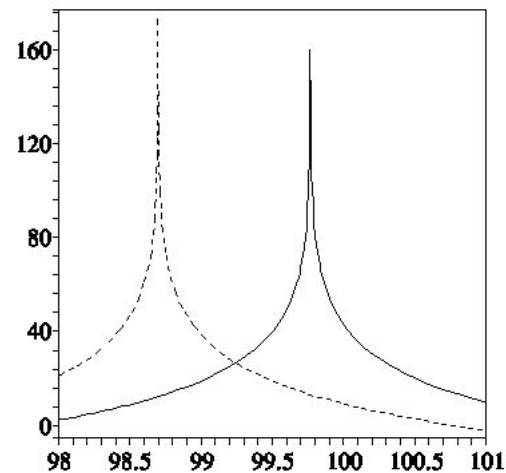
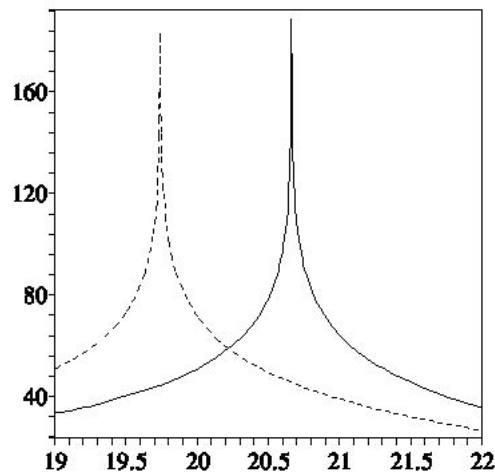
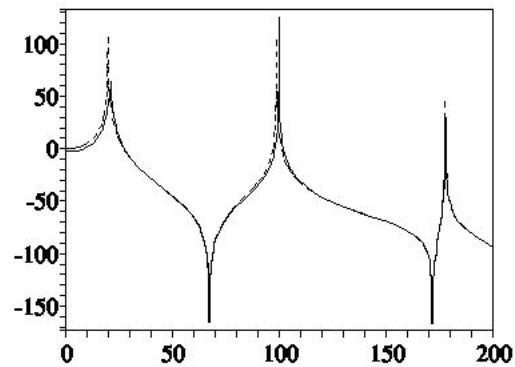
$$= \frac{1}{2} \frac{d}{D^{eff} + d} \frac{1}{A} \left(\int_A \left[\int_A \hat{p}(\mathbf{x}) (G_1^\alpha(\mathbf{x}, \xi) + G_2^\alpha(\mathbf{x}, \xi)) dA(\mathbf{x}) \right] dA(\xi) \right)$$

PART II: Numerical example

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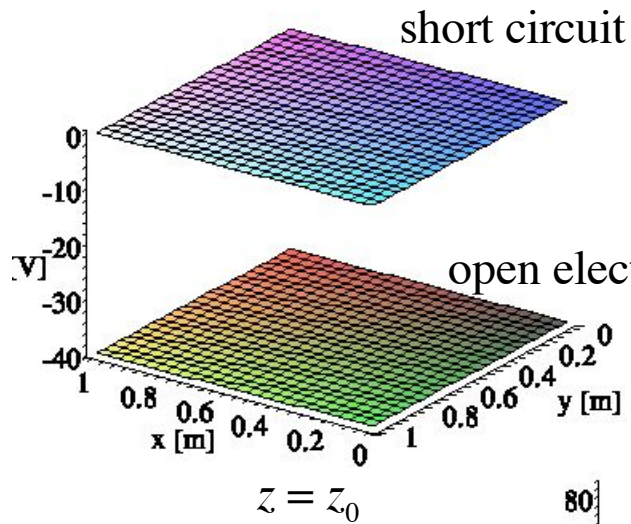
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Dynamic magnification factor for different ranges of α

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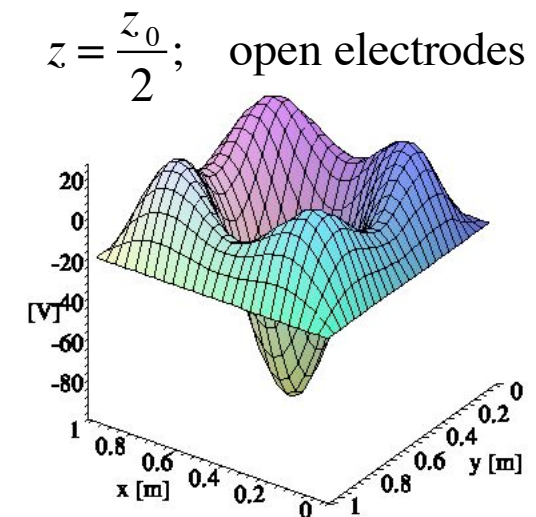
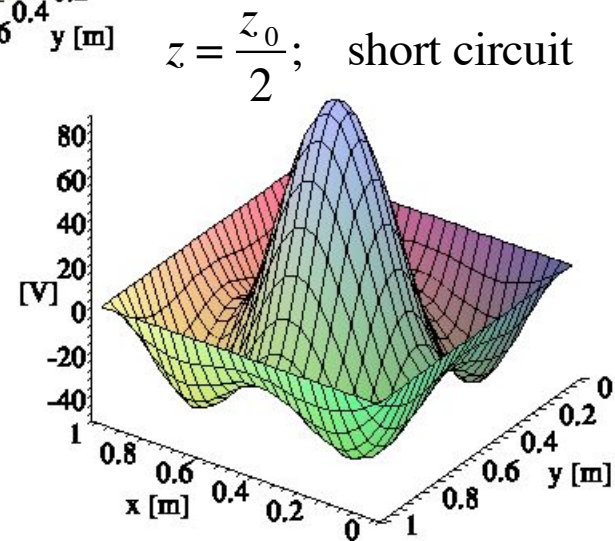
PART II: Numerical example



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Electric potential distribution for $\alpha = 99.2$

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PART III: Governing equations

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$$A: \operatorname{div}(\operatorname{div}\mathbf{M}) + p_z = P\ddot{w}_0$$

$$C: \left[\operatorname{div}\mathbf{M} \cdot \mathbf{n} + \nabla(\mathbf{M}\mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} - \bar{q} \right] \delta w_0 = 0, \quad (\mathbf{M}\mathbf{n} \cdot \mathbf{n} - \bar{m})(\nabla \delta w_0 \cdot \mathbf{n}) = 0$$

$$P: (\mathbf{M}\mathbf{n} \cdot \mathbf{s}) \delta w_0 \Big|_{P^+}^{P^-} = 0$$

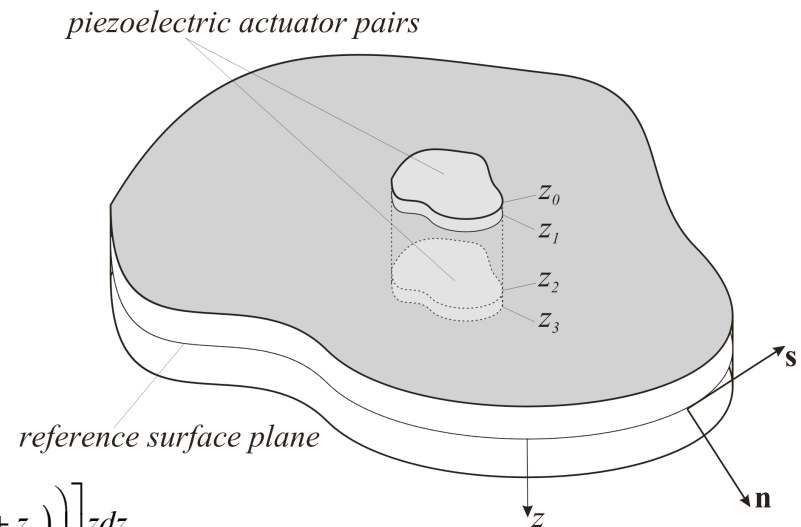
with patch: $\mathbf{M} = \mathbf{D}^{eff} : \boldsymbol{\kappa} - \mathbf{M}^{*,eff}$, $P = P_0 + \Delta P$

without patch: $\mathbf{M} = \mathbf{D}_0 : \boldsymbol{\kappa}$, $P = P_0$

$$P_0 = \int_{z_0}^{z_1} \rho^{substrate} dz, \quad \Delta P = 2 \int_{z_0}^{z_1} \rho^{patch} dz$$

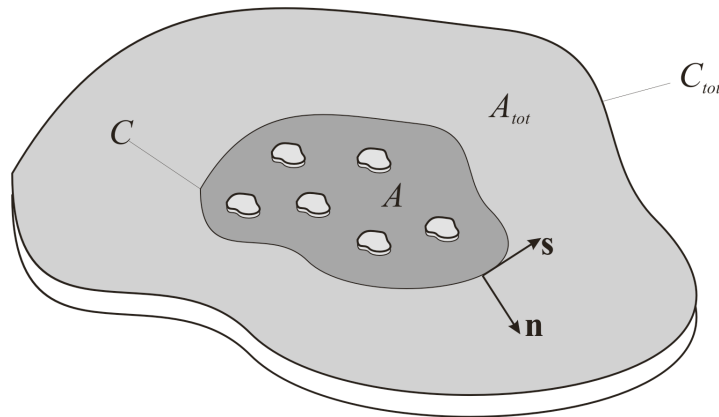
$$\mathbf{M}^{*,eff} = \int_{z_0}^{z_1} \bar{\mathbf{e}} \frac{V^u}{h_p} z dz + \int_{z_2}^{z_3} \bar{\mathbf{e}} \frac{V^l}{h_p} z dz = \int_{z_0}^{z_1} \bar{\mathbf{e}} \frac{2V^u}{h_p} z dz, \quad \mathbf{D}_0 = \int_{z_1}^{z_2} \mathbf{C}^{substrate} z^2 dz$$

$$\Delta \mathbf{D} = \int_{z_0}^{z_1} \left[\mathbf{C}^{patch} z + \frac{\bar{\mathbf{e}} \otimes \bar{\mathbf{e}}}{\eta} \left(z - \frac{1}{2}(z_0 + z_1) \right) \right] z dz + \int_{z_2}^{z_3} \left[\mathbf{C}^{patch} z + \frac{\bar{\mathbf{e}} \otimes \bar{\mathbf{e}}}{\eta} \left(z - \frac{1}{2}(z_2 + z_3) \right) \right] z dz$$



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PART III: Exact solution



$$C : \begin{cases} \operatorname{div} \mathbf{M} \cdot \mathbf{n} + \nabla (\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} = \bar{q}, & \mathbf{M} \mathbf{n} \cdot \mathbf{n} = \bar{m} \\ w_0 = \bar{w}_0, & \mathbf{M} \mathbf{n} \cdot \mathbf{n} = \bar{m} \\ \operatorname{div} \mathbf{M} \cdot \mathbf{n} + \nabla (\mathbf{M} \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} = \bar{q}, & \nabla w_0 \cdot \mathbf{n} = \bar{\psi} \\ w_0 = \bar{w}_0, & \nabla w_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$

$$\int_0^t \int_A [\mathbf{M}^p + \mathbf{M}^{*,eff}] (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA dt =$$

$$\int_0^t \left[\int_A p_z^d (\mathbf{x}, t - \tau) \tilde{w}_0 (\mathbf{x}, t) dA + \int_{C_q} \bar{q}^d (\mathbf{x}, t - \tau) \tilde{w}_0 (\mathbf{x}, t) dC - \int_{C_m} \bar{m}^d (\mathbf{x}, t - \tau) (\nabla \tilde{w}_0 (\mathbf{x}, t) \cdot \mathbf{n}) dC \right] dt$$

If $\mathbf{M}^p + \mathbf{M}^* = \mathbf{0}$, then $w_0 = \tilde{w}_0$.

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PART III: Exact solution

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$$A: \operatorname{div}(\operatorname{div}\mathbf{M}^p) + \tilde{p}_z = 0$$

$$C: \begin{cases} \operatorname{div}\mathbf{M}^p \cdot \mathbf{n} + \nabla(\mathbf{M}^p \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} = \tilde{q}, & \mathbf{M}^p \mathbf{n} \cdot \mathbf{n} = \tilde{m} \\ \mathbf{M}^p \mathbf{n} \cdot \mathbf{n} = \tilde{m} \\ \operatorname{div}\mathbf{M}^p \cdot \mathbf{n} + \nabla(\mathbf{M}^p \mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s} = \tilde{q} \\ \text{no boundary conditions to be satisfied} \end{cases}$$

$$A: \tilde{p}_z = p_z - \bar{P}\ddot{w}_0 + \operatorname{div}(\operatorname{div}\check{\mathbf{M}}), \quad \check{\mathbf{M}} = \bar{\mathbf{D}} : \check{\boldsymbol{\kappa}}, \quad \check{\boldsymbol{\kappa}} = -\frac{1}{2} \left[\nabla\nabla\check{w}_0 + (\nabla\nabla\check{w}_0)^T \right]$$

$$C: \tilde{q} = \bar{q} - \operatorname{div}\check{\mathbf{M}} \cdot \mathbf{n} - \nabla(\check{\mathbf{M}}\mathbf{n} \cdot \mathbf{s}) \cdot \mathbf{s}, \quad \tilde{m} = \bar{m} - \check{\mathbf{M}}\mathbf{n} \cdot \mathbf{n}$$

$$A: \nabla\nabla\check{w}_0 = \mathbf{0}, \quad C: \begin{cases} \text{no conditions to be satisfied} \\ \check{w}_0 = \bar{w}_0 \\ \nabla\check{w}_0 \cdot \mathbf{n} = \bar{\psi} \\ \check{w}_0 = \bar{w}_0, \quad \nabla\check{w}_0 \cdot \mathbf{n} = \bar{\psi} \end{cases}$$

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PART III: *Approximate solution*



- $i = 1, \dots, n$ patches within A
- Each patch is located within A_i^{sub} with $A_i^{sub} \subseteq A_i$ and $\bigcap_{i=1}^n A_i = A$
- Piezoelectric actuation at patch i : $\mathbf{M}_i^{*,eff}(t)$

$$\int_0^t \int_A \left[\mathbf{M}^P + \mathbf{M}^{*,eff} \right] (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA dt =$$
$$= \int_0^t \left[\sum_{i=1}^n \left(\mathbf{M}_i^{*,eff} (t - \tau) : \int_{A_i^{sub}} \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right) + \int_A \mathbf{M}^P (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right] dt$$

PART III: ***Approximate solution; direct method***

$$\int_0^t \int_A [\mathbf{M}^P + \mathbf{M}^{*,eff}] (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA dt =$$
$$= \int_0^t \left[\sum_{i=1}^n \left(\mathbf{M}_i^{*,eff} (t - \tau) : \int_{A_i^{sub}} \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right) + \int_A \mathbf{M}^P (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right] dt$$

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$$\mathbf{M}_i^{*,eff} (t) = - \frac{1}{A_i^{sub}} \int_{A_i} \mathbf{M}^P (\mathbf{x}, t) dA$$

- High number of piezoelectric patches possible
- Easy to use

***Control of Bending Vibrations within Subdomains
of Thin Plates by Piezoelectric Actuation***

PART III: Approximate solution; indirect method



$$\int_0^t \int_A \left[\mathbf{M}^P + \mathbf{M}^{*,eff} \right] (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA dt =$$

$$= \int_0^t \left[\sum_{i=1}^n \left(\mathbf{M}_i^{*,eff} (t - \tau) : \int_{A_i^{sub}} \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right) + \int_A \mathbf{M}^P (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right] dt$$

- Special dummy loading cases are considered; for each case

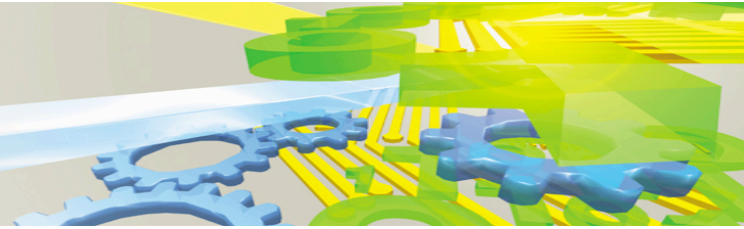
$$\int_0^t \sum_{i=1}^n \left(\mathbf{M}_i^{*,eff} (t - \tau) : \int_{A_i^{sub}} \boldsymbol{\kappa}^{d,j} (\mathbf{x}, t) dA \right) dt = - \int_0^t \int_A \mathbf{M}^P (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^{d,j} (\mathbf{x}, t) dA dt$$

- n dummy loading cases required

- Mechanical interpretation
of quality of approximate solution

$$\left\{ \begin{array}{l} \bar{m}^{d,j} = 0, \quad \bar{q}^{d,j} = 0, \quad p_z^{d,j} (\mathbf{x}, t) = f(t) \\ \text{then } \int_0^t f(t - \tau) \int_A \tilde{w}_0 (\mathbf{x}, t) dA dt = 0 \end{array} \right.$$

PART III: Approximate solution; combined method



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$$\int_0^t \int_A [\mathbf{M}^P + \mathbf{M}^{*,eff}] (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA dt =$$

$$= \int_0^t \left[\sum_{i=1}^n \left(\mathbf{M}_i^{*,eff} (t - \tau) : \int_{A_i^{sub}} \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right) + \int_A \mathbf{M}^P (\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^d (\mathbf{x}, t) dA \right] dt$$

- Statically admissible moment and self-moment separable:

$$\mathbf{M}^P (\mathbf{x}, t) = \sum_{j=1}^m \mathbf{M}_j^P (\mathbf{x}) f_j (t), \quad \mathbf{M}^{*,eff} (\mathbf{x}, t) = \sum_{j=1}^m \mathbf{M}_j^{*,eff} (\mathbf{x}) g_j (t)$$

- $\mathbf{M}_j^P (\mathbf{x})$'s account for space wise distributions of: $\tilde{p}_z, \tilde{m}, \tilde{q}$

- Patch self-moment are taken as: $\mathbf{M}_i^{*,eff} (t) = \sum_{j=1}^m \mathbf{M}_{ij}^{*,eff} g_j (t)$ with: $\mathbf{M}_{ij}^{*,eff} = -\frac{1}{A_i^{sub}} \int_{A_i} \mathbf{M}_j^P (\mathbf{x}) dA$

*Control of Bending Vibrations within Subdomains
of Thin Plates by Piezoelectric Actuation*



PART III: ***Approximate solution; combined method***

- m equations for m unknown time variations $g_j(t)$

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$$\int_0^t g_j(t - \tau) \sum_{i=1}^n \left(\mathbf{M}_{ij}^{*,eff} : \int_{A_i^{sub}} \boldsymbol{\kappa}^{d,j}(\mathbf{x}, t) dA \right) dt = - \int_0^t \int_A \mathbf{M}^p(\mathbf{x}, t - \tau) : \boldsymbol{\kappa}^{d,j}(\mathbf{x}, t) dA dt$$

by using m dummy loading cases:

$$p_z^{d,j}(\mathbf{x}, t) = \tilde{p}_z(\mathbf{x}, t), \quad \bar{m}^{d,j}(\mathbf{x}, t) = \tilde{m}(\mathbf{x}, t), \quad \bar{q}^{d,j}(\mathbf{x}, t) = \tilde{q}(\mathbf{x}, t)$$

- High number of piezoelectric patches possible
- Mechanical interpretation of quality of approximate solution

***Control of Bending Vibrations within Subdomains
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PART III: ***Numerical example***

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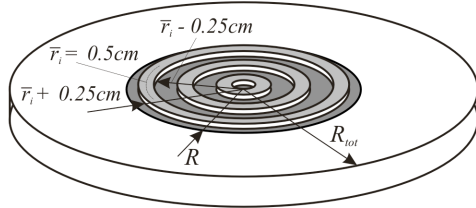
- Clamped circular Plate: $R_{tot} = 0.2m$ and $h = 3mm$
- Natural frequencies of aluminium plate: $f_1 = 182.0Hz, f_2 = 708.7Hz, f_3 = 1588Hz$
- Control deflection within $r \leq R = 3cm$ to be $w_0(r, t) = \bar{w}_0 f(t)$
- Statically admissible moment: $r \leq R$: $div(div \mathbf{M}^p) + p_z - \bar{\rho}^{(0)} \bar{w}_0 \ddot{f}(t) = 0$, $r = R$: $\mathbf{M}^p \mathbf{n} \cdot \mathbf{n} = \bar{m}$
- Force loading: $r \leq R$: $p_z(r, t) = p_0 e^{i\omega t}$, $p_0 = 100N / m^2$, $\omega = 2\pi f = 2\pi(100, 500, 1000) s^{-1}$
- Statically admissible moment: $r \leq R$: $\Delta M^p + p_0 + \omega^2 \bar{\rho}^{(0)} \bar{w}_0 = 0$, $r = R$: $M^p = \bar{m}$

***Control of Bending Vibrations within Subdomains
of Thin Plates by Piezoelectric Actuation***

PART III:

Comparison direct and indirect method

■ controlled sub - domain
 ◡ piezoelectric rings



- 3 PZT-5A concentric patches: $h_p = 0.5\text{mm}$
 $r \in [\bar{r}_i \pm 0.25\text{cm}]$, $\bar{r}_i = (0.5, 1.5, 2.5)\text{cm}$

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- Direct method: $M_i^{*,eff} \int_{r_i-0.0025}^{r_i+0.0025} 2\pi r dr = - \int_{r_i-0.005}^{r_i+0.005} M^P(r) 2\pi r dr$

- Indirect method: $\sum_{i=1}^3 \left(M_i^{*,eff} \int_{r_i-0.0025}^{r_i+0.0025} \Delta w_0^{d,j}(r) 2\pi r dr \right) + \int_0^R M^P(r) \Delta w_0^{d,j}(r) 2\pi r dr = 0$

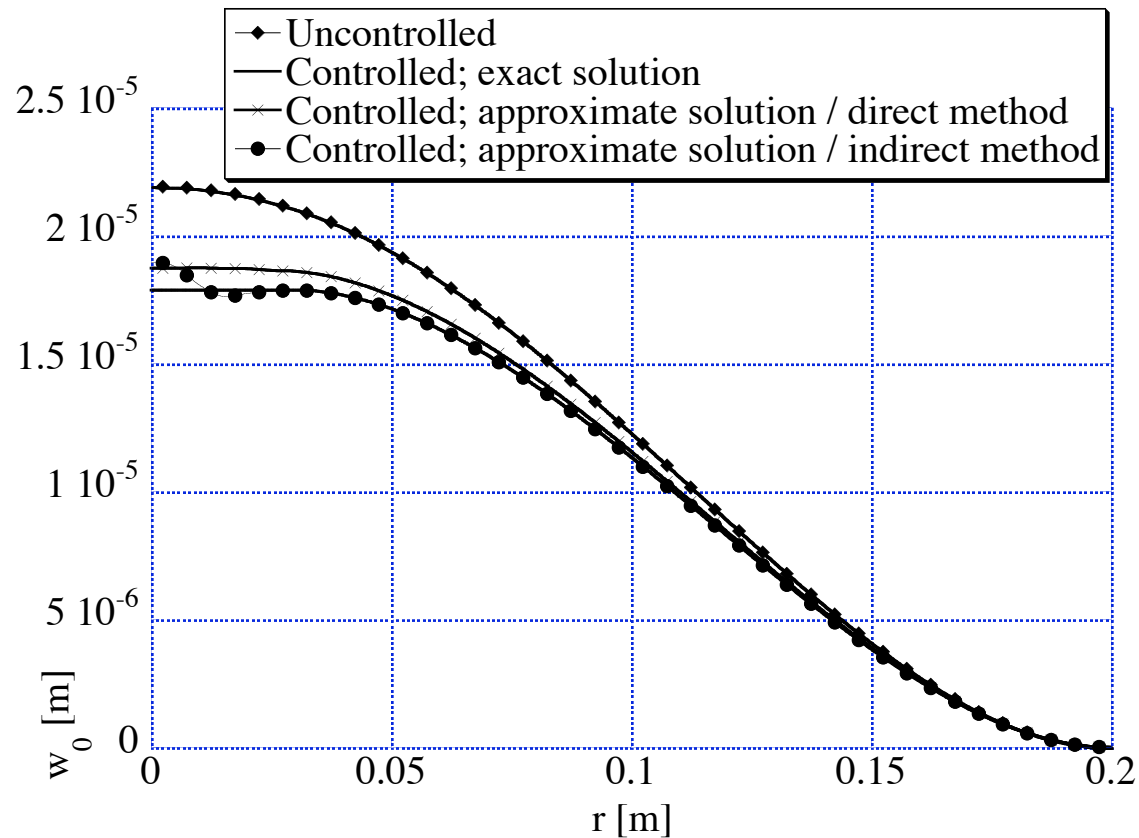
- Dummy loadings: $p_z^{d,1}(r,t) = 1e^{i\omega t}$, $p_z^{d,2}(r,t) = (\bar{P} / P_0)e^{i\omega t}$, $\bar{m}^{d,2} = 1e^{i\omega t}$

$$\text{then } \int_0^R \tilde{w}_0(r) 2\pi r dr = 0 \text{ and } \int_0^R \frac{\bar{P}}{P_0} \tilde{w}_0(r) 2\pi r dr = 0 \text{ and } \left[\frac{\partial \tilde{w}_0(r)}{\partial r} \right]_{r=R} = 0.$$

*Control of Bending Vibrations within Subdomains
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PART III:

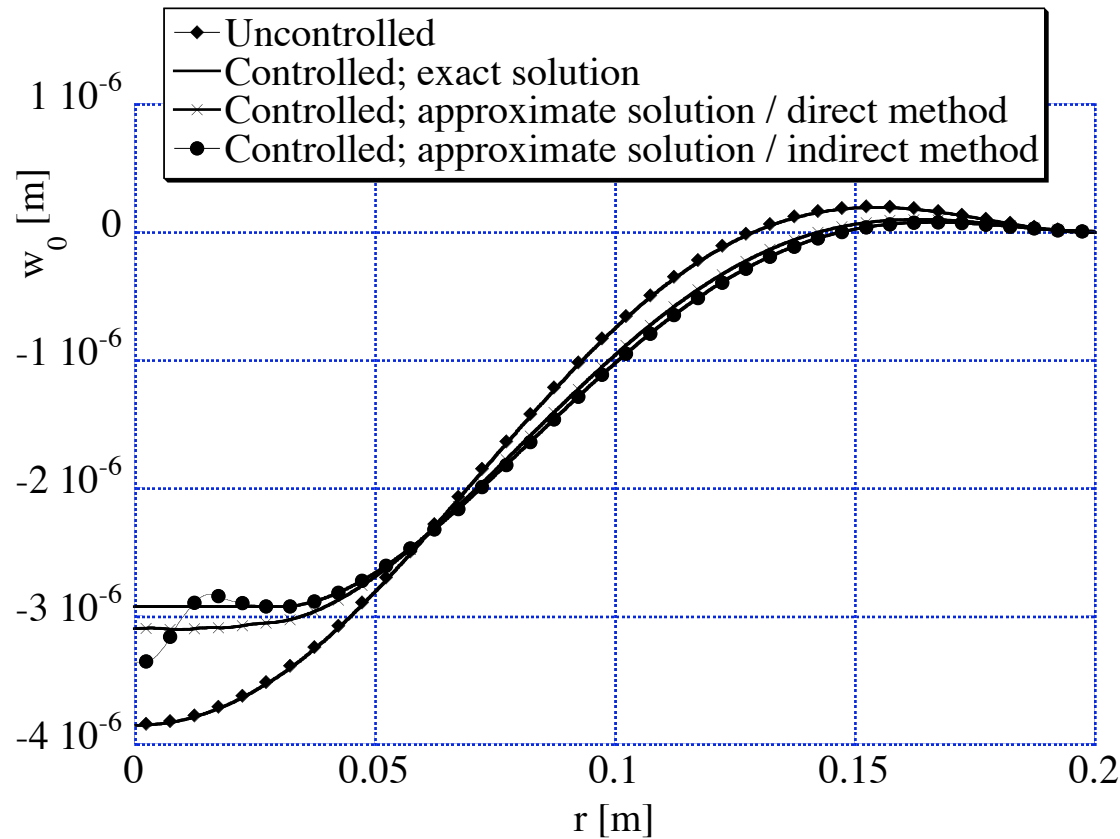
Comparison direct and indirect method



$$\omega = 2\pi 100 \text{ s}^{-1}$$

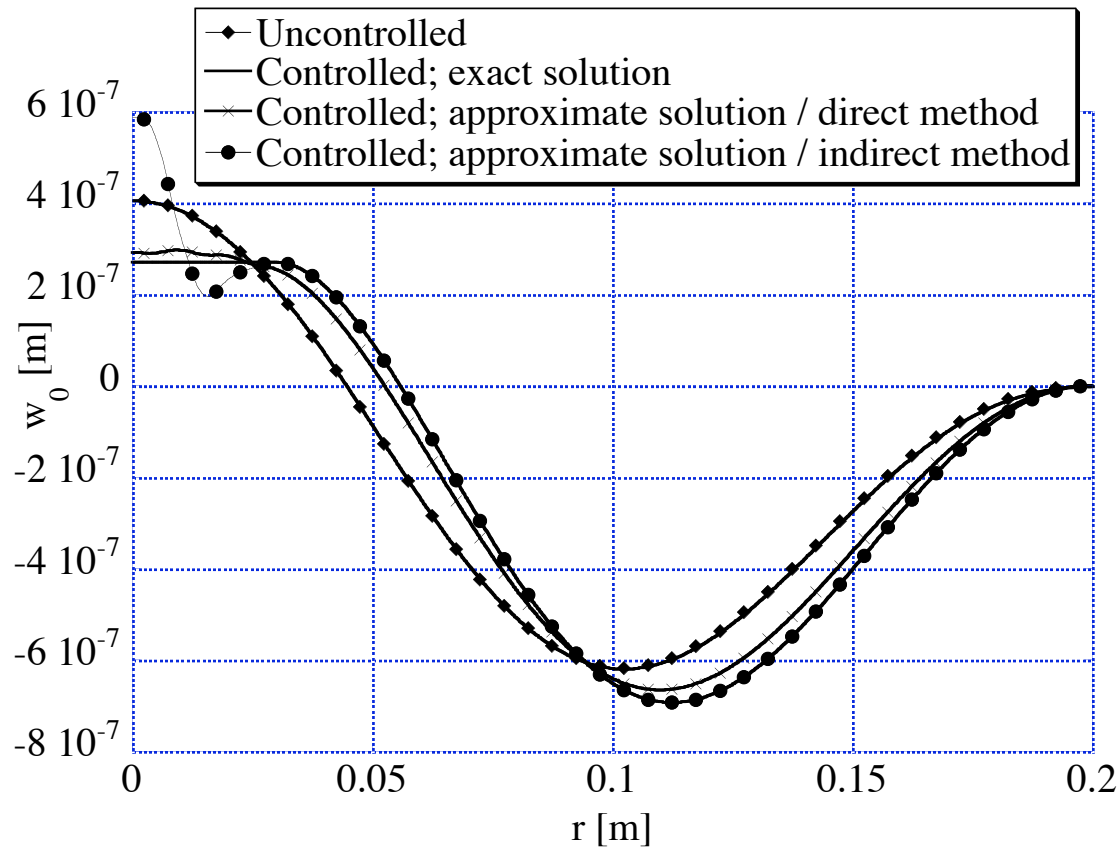
PART III:

Comparison direct and indirect method



$$\omega = 2\pi 500 \text{ s}^{-1}$$

PART III: Comparison direct and indirect method



$$\omega = 2\pi 1000 s^{-1}$$

PART III: Combined method

- 6 PZT-5A concentric patches: $h_p = 0.5 \text{ mm}$

$$r \in [\bar{r}_i \pm 0.125 \text{ cm}], \quad \bar{r}_i = (0.25, 0.75, 1.25, 1.75, 2.25, 2.75) \text{ cm}$$

- Combined method: $M_{ij}^{*,eff} \int_{r_i-0.00125}^{r_i+0.00125} 2\pi r dr = - \int_{r_i-0.0025}^{r_i+0.0025} M_j^P(r) 2\pi r dr$

$$M_1^P \leftrightarrow p_0, \quad M_2^P \leftrightarrow \bar{P} / P_0, \quad M_3^P \leftrightarrow \bar{m} \text{ at } r = R$$

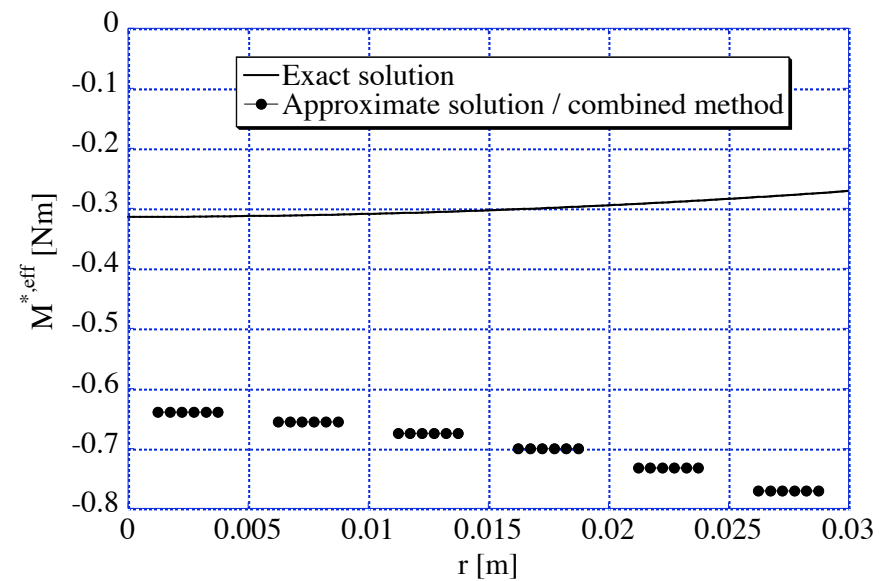
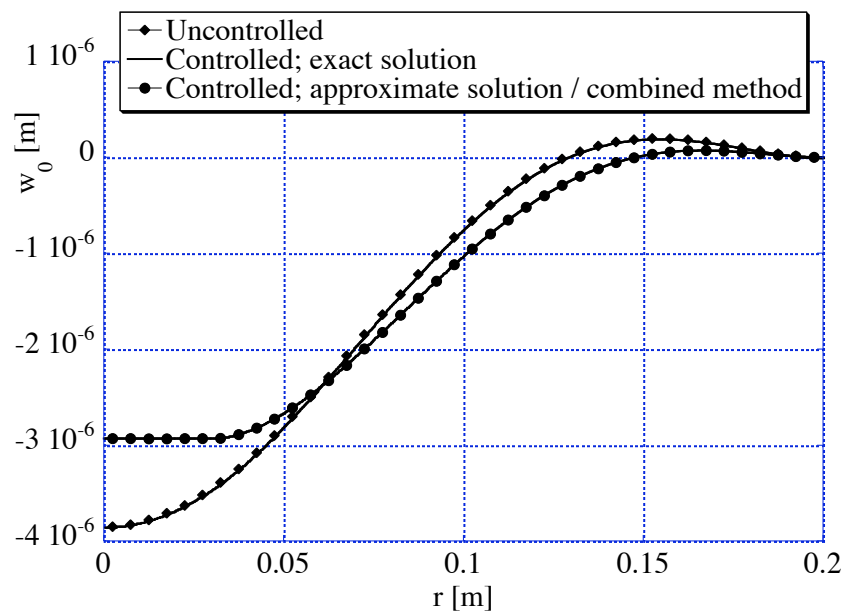
- Self-moment at each concentric ring: $M_i^{*,eff} e^{i\omega t} = \sum_{j=1}^3 M_{ij}^{*,eff} K_j e^{i\omega t}$

$$\sum_{i=1}^6 \left(\sum_{j=1}^3 M_{ij}^{*,eff} K_j \right) \int_{r_i-0.00125}^{r_i+0.00125} \Delta w_0^{d,k}(r) 2\pi r dr = - \int_0^R M^P(r) \Delta w_0^{d,k}(r) 2\pi r dr$$

- Dummy loadings: $p_z^{d,1}(r,t) = 1 / (2\pi \bar{r}) e^{i\omega t}$, $p_z^{d,2}(r,t) = (\bar{P} / P_0) e^{i\omega t}$, $\bar{m}^{d,2} = 1 e^{i\omega t}$

$$\text{then } \int_0^R \tilde{w}_0(r) r / \bar{r} dr = 0 \quad \text{and} \quad \int_0^R \frac{\bar{P}}{P_0} \tilde{w}_0(r) 2\pi r dr = 0 \quad \text{and} \quad \left[\frac{\partial \tilde{w}_0(r)}{\partial r} \right]_{r=R} = 0.$$

PART III: Combined method



$$\omega = 2\pi 500 s^{-1}$$

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PART III: Control of alternative subdomain

- $0.16m = R_{sub} \leq r \leq R_{tot} = 0.2m$: 4 PZT-5A concentric patches
 $r \in [\bar{r}_i \pm 0.25cm]$, $\bar{r}_i = (0.165, 0.175, 0.185, 0.195)m$

$$R_{sub} \leq r \leq R_{tot} : \Delta M^P + p_0 = 0, \quad r = R_{sub} : M^P = \bar{m} \quad \text{and} \quad \frac{\partial M^P}{\partial r} = \bar{q}$$

- Combined method: $M_{ij}^{*,eff} \int_{r_i-0.0025}^{r_i+0.0025} 2\pi r dr = - \int_{r_i-0.005}^{r_i+0.005} M_j^P(r) 2\pi r dr$

$$M_1^P \leftrightarrow p_0, \quad M_2^P \leftrightarrow \bar{q} \text{ at } r = R, \quad M_3^P \leftrightarrow \bar{m} \text{ at } r = R$$

- Self-moment at each concentric ring: $M_i^{*,eff} e^{i\omega t} = \sum_{j=1}^3 M_{ij}^{*,eff} K_j e^{i\omega t}$

PART III: Control of alternative subdomain

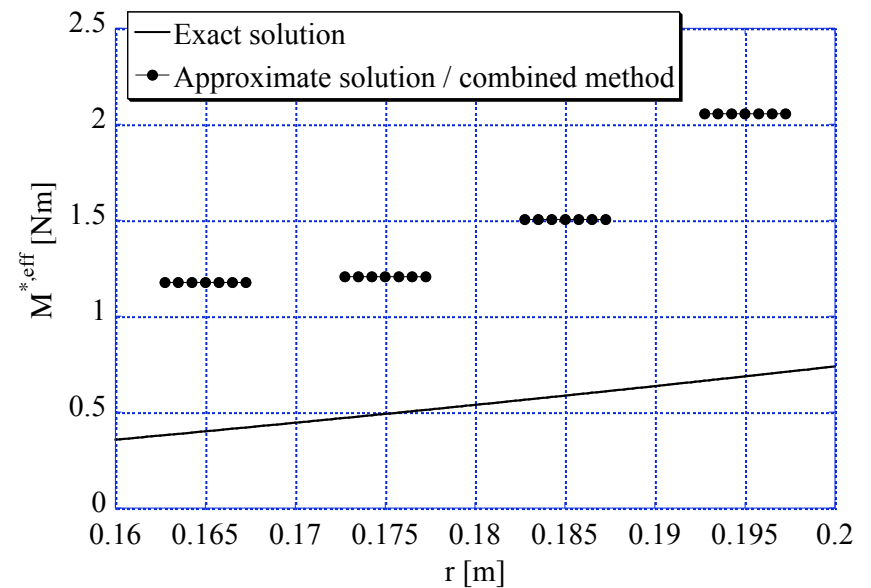
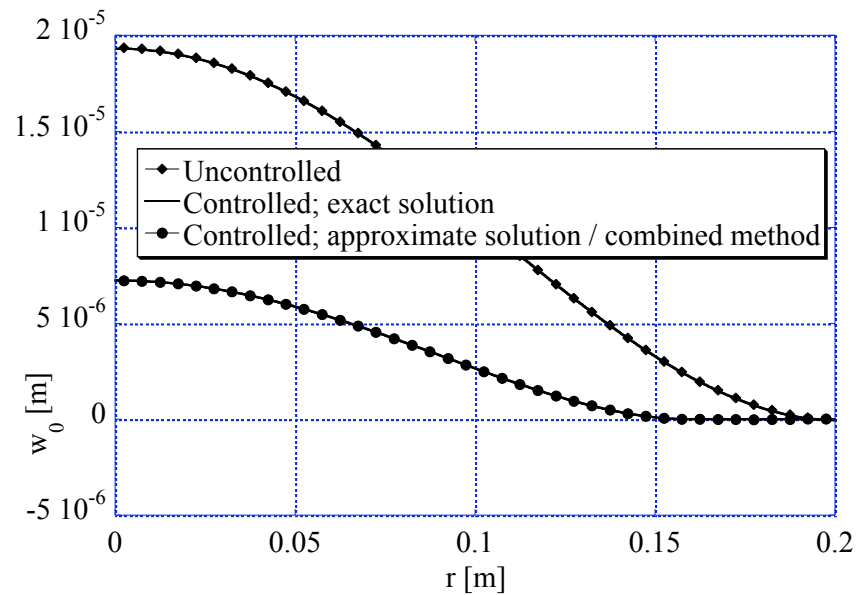


$$\sum_{i=1}^4 \left(\sum_{j=1}^3 M_{ij}^{*,eff} K_j \right) \int_{r_i-0.0025}^{r_i+0.0025} \Delta w_0^{d,k}(r) 2\pi r dr = - \int_{R_{sub}}^{R_{tot}} M^p(r) \Delta w_0^{d,k}(r) 2\pi r dr$$

- Dummy loadings: $p_z^{d,1}(r,t) = 1 / (2\pi\bar{r}) e^{i\omega t}$, $\bar{q}^{d,2} = 1 e^{i\omega t}$, $\bar{m}^{d,2} = 1 e^{i\omega t}$

$$\text{then } \int_{R_{sub}}^{R_{tot}} \tilde{w}_0(r) \frac{2\pi r}{2\pi\bar{r}} dr = 0 \text{ and } [\tilde{w}_0(r)]_{r=R_{sub}} = 0 \text{ and } \left[\frac{\partial \tilde{w}_0(r)}{\partial r} \right]_{r=R} = 0.$$

PART III: Control of alternative subdomain



$$\omega = 2\pi 100 s^{-1}$$

Control of Bending Vibrations within Subdomains
of Thin Plates by Piezoelectric Actuation